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# Direct Power Control of DFIG Wind Turbine Systems Based on an Intelligent Proportional-Integral Sliding Mode Control

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19 *Keywords:* Wind turbine system; Model-free control; Sliding mode control

20

# 21 **1. INTRODUCTION**

22 As a consequence of population expansion and increasing environmental issues, the 23 demand for renewable energy generation systems keeps growing. As a green and 24 clean energy, wind turbine systems have been paid considerable attention and their 25 proportion in nationwide energy production will rise in the next decade according to 26 the Global Wind Energy Council report [1, 2]. However, random wind fluctuations 27 and wind turbine nonlinearity are major difficulties for exploiting renewable energy 28 with a high efficiency. The nonlinear characteristics of a wind turbine system can be 29 classified as electrical and mechanical nonlinearities. While the former are related to 30 the generators and its uncertain parameters; the latter are related to the drive train and 31 wind wheels for instance. Considering both electrical and mechanical nonlinearities, designing an efficient wind turbine controller is a challenging problem. 32

Wind turbine systems are high-order nonlinear systems. The doubly fed induction generator (DFIG) is widely utilized on the multi-MW wind turbines because of its low cost and small size. Their nonlinear characteristics are not only reflected in the DFIG model, but also in the aerodynamic and drive-train models. With large power wind

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#### Fig. 1. The DFIG wind turbine system

turbine systems developing and blade diameter increasing, its nonlinear feature will
be reinforced, and will influence directly the output performance of wind turbine
systems.

42 Modeling and control of wind turbine systems has been a vivid research topic in the 43 past decade [3]. A controller can optimize the power production of a DFIG in many 44 ways. For speed and torque or power control of DFIGs, there are vector control, direct 45 torque control and direct power control [4].

46 In low wind speed region (between cut-in speed and rating speed ), most of reported 47 methods in the literature aim at tracking the maximum power point (MPP) of DFIGs. 48 In reference [5], a direct power control strategy based on proportional-integral (PI) 49 controller has been developed for DFIGs. Even though this method ensures an 50 input-to-state closed-loop stability, it does not take electrical nonlinearities into 51 account. Considering electrical nonlinearities which are originated in DFIG parameter 52 uncertainties both in resistance and inductance, a sliding mode control approach has 53 been proposed for regulation of the active and reactive power in [6]. In order to 54 circumvent external uncertainty sources such as wind turbulences, a robust fuzzy 55 controller and a fuzzy logic controller for direct power regulation are designed in [7] 56 and [8]. Another type of controller is introduced in [9, 10] for MPP tracking. They use 57 a radial-basis function neural network controller which focuses only on the nonlinear 58 aerodynamic model and neglects the electrical torque equation. The same nonlinear 59 aspects of the aerodynamic model are also taken into account in [11] using the same 60 family of controllers as in [6], i.e. sliding mode controllers. Also in [12], a 61 discrete-time sliding mode approach is introduced for variable speed wind turbine 62 system.

In order to improve the efficiency of MPP tracking, a novel controller relying on an
intelligent proportional-integral-derivative (iPID) control is proposed in [13], it has
been proved to produce efficient control for a variety of systems such as a quadrotor
vehicle [14], DC/DC converters, and motors [15]. This approach uses an observer

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which is based on algebraic techniques to estimate the unknown dynamics [16].
However, this algebraic based iPID control cannot ensure the trajectory tracking error
to tend to zero rapidly. In addition, its estimation performances are significantly
degraded by measurement noises [17].

71 In order to overcome the aforementioned difficulties, this paper presents an extended 72 state observer based intelligent proportional-integral sliding mode control (iPISMC) 73 to perform direct power control of DFIG Wind turbine systems. The extended state 74 observer (ESO) is integrated into an intelligent proportional-integral (iPI) to estimate 75 the unknown uncertain dynamics of the system. An acceptable performance can be 76 ensured when the unknown dynamic is bounded and the parameters of ESO observer 77 are carefully selected [18]. Unfortunately, there always remains a non-null estimation 78 error if the ESO observer is not well selected. Concerning this estimation error, an 79 auxiliary sliding mode controller is added to the ESO based iPI control. The full 80 control strategy that we propose will be referred to as iPISMC. With application of the 81 Lyapunov stability theory, we prove the stability of the proposed iPISMC control. 82 Using simulations generated by the FAST/simulink platform, we show that the 83 proposed controller is robust to random wind inputs and parameter variations. The 84 experiments also demonstrate that iPISMC outperforms PI and iPI controllers in terms 85 of average power production. Note that our goal in this paper is not to prove that 86 iPISMC outperforms any other controller but only to validate that for given proportional and integral gain values, it should be preferred to PI or iPI controllers. 87

The paper is organized as follows. In section II, wind turbine system modeling and the basic principle of vector control for DFIG will be briefly presented.

In Section III, an intelligent proportional and integral sliding mode controller is
designed. Some simulation results are shown in Section IV assessing the quality of
iPISMC in terms of power production and response time. At last, section V concludes
the paper.

# 94 **2.** Wind turbine system modeling and vector control

95 The DFIG based wind turbine system which is illustrated in Fig. 1, is mainly composed of the following three components: the aerodynamic subsystem, the DFIG 96 97 subsystem, and the drive-train subsystem. From Fig. 1, one notes that the general 98 control strategy is based on two loops: the inner loop which regulates the rotor current, 99 and the outer loop which is applied to track the maximum power point. The aerodynamic and gearbox subsystems will be simulated by the FAST platform which 100 101 is developed by the National Renewable Energy Laboratory (NREL). In this paper, we 102 focus on the maximum power point tracking.

# 103 **2.1 Aerodynamic subsystem**

104 Usually, the approximate values of aerodynamic power  $P_a$  and torque  $T_a$  are given by 105 the following equations:

106  

$$\begin{cases}
P_{a} = \frac{1}{2} \rho \pi R^{2} v^{3} C_{p}(\lambda, \beta) \\
T_{a} = \frac{1}{2} \rho \pi R^{3} v^{2} C_{p}(\lambda, \beta) / \lambda
\end{cases}$$
(1)

107 where  $\lambda$  is tip speed ratio and we have  $\lambda = \omega_m R/v$ . *R* is the blade radius, *v* is the wind 108 speed,  $\rho$  is the air density,  $\beta$  is the pitch angle and  $C_p$  is the power coefficient.  $\omega_m$  is 109 the rotor speed.

In a variable pitch and variable speed system, by changing pitch angle, when wind flows through wind turbine, its output power will be varying with respect to rotor speed and pitch angle. In order to obtain more energy under a given pitch angle value, we can set  $\lambda$  as an optimal value so that the power coefficient can reach a maximum value. Therefore, one typical method for tracking the maximum power is to maintain the tip speed ratio constant by measuring the wind speed and rotor speed [19, 20].

#### 116 **2.2 DFIG subsystem**

117 The induction generator can be written in *dq* arbitrary reference frame as follows [21]:

118  

$$\begin{cases}
 u_{sd} = R_{s}i_{sd} - \omega_{s}i_{sd} + \dot{\psi}_{sd} \\
 u_{sq} = R_{s}i_{sq} + \omega_{s}i_{sq} + \dot{\psi}_{sq} \\
 u_{rd} = R_{s}i_{rd} - \omega_{r}i_{rd} + \dot{\psi}_{rd} \\
 u_{rq} = R_{s}i_{rq} + \omega_{r}i_{rq} + \dot{\psi}_{rq}
 \end{bmatrix}$$
(2)

119 where  $\dot{\psi}_{sd}, \dot{\psi}_{sq}, \dot{\psi}_{rd}, \dot{\psi}_{rq}$  are the derivatives of fluxes  $\psi_{sd}, \psi_{sq}, \psi_{rd}, \psi_{rq}$ , respectively.

#### 120 **2.3 Drive train subsystem**

In the DFIG subsystem, the aerodynamic torque is transferred to generator side by the
gearbox. The drive train subsystem can be simplified regardless of friction loss. It can
be written as follows [22]:

124

$$T_a - T_e = J\dot{\omega}_m + B\omega_m, \qquad (3)$$

(2)

125 where  $T_a$  is the equivalent aerodynamic torque, J is the equivalent moment of inertia 126 and B is damping factor.  $T_e$  is the electromagnetic torque.

#### 127 **2.4 Vector Control Strategy of the DFIG based wind turbine system**

In order to regulate the power of the DFIG based wind turbine system, a common 128 129 method is to utilize a vector control by flux orientation, such as stator flux orientation 130 (SFO) [23], stator voltage orientation (SVO). By dq coordinate transformation, lots of methods can be developed and their controllers are powerful in different aspects. In 131 fact, the differences between these methods are on the control strategy and measured 132 133 variables [24]. Here, the chosen method for DFIG power regulation is SFO. In steady 134 conditions, voltage and frequency are approximately constant and one has the 135 following relationships

136 
$$\begin{cases} \psi_{sd} = \psi, \psi_{sq} = 0\\ u_{sq} = \omega_1 \psi, u_{sd} = 0 \end{cases}$$
 (4)

By making substitutions in equation (2) using (4), the equivalent rotor dynamicmodels can be derived as

139
$$\begin{cases} \dot{i}_{rd} = -(R_r i_{rd} - \omega_r i_{rq}) / \delta L_r + u_{rd} / \delta L_r \\ \dot{i}_{rq} = -(R_r i_{rq} + \omega_r i_{rd}) / \delta L_r + u_{rq} / \delta L_r - \omega_r (L_m / L_s) \psi, \end{cases}$$
(5)

140 where  $\delta = 1 - L_m^2 / (L_r L_s)$  is a leak coefficient and  $\omega_r$  is the rotor electrical speed in the

141 synchronous reference frame. Equation (5) indicates that the current of d or q axis is 142 not strictly independent of d or q voltage in rotor side.

143 To perform the decoupled control and achieve high-performance, two different offset 144 voltages will be added to *dq* voltages as illustrated in Fig. 2.

145 The offset voltages can be calculated as

146 
$$\begin{cases} \Delta u_{rd} = -\omega_r \delta L_r i_{rq} \\ \Delta u_{rq} = \omega_r \delta L_r i_{rd} + \omega_r (L_m / L_s) \psi \end{cases}$$
(6)

147 With the offset voltages, the model of DFIG will be simplified and decoupled. The 148 parameter of inner-loop proportional-Integral controller can be designed according to pole placement. This scheme is used assuming that the stator voltage is fixed and that 149 150 the compensated voltage can annihilate completely the offset. However, in practice, 151 the performance of a PI controller designed by that method depends on its invariance 152 with respect to system parameters whose values must be known beforehand. Besides, the measurement noise, flux saturation and other nonlinear factors will also increase 153 154 power error.

The structure of active and reactive power control is shown in Fig. 3. It can be treated as inner loop and outer loop control. The main objective for the outer loop is to regulate active and reactive powers. The output of the outer loop will serve as a reference for the inner loop.



159 160

Fig. 2. DFIG model in *dq* reference frame

#### 161 **3.** Intelligent proportional-integral Sliding Mode Control

162 In this part, the basic principle of iPI and iPISMC is introduced. The stability of 163 iPISMC in closed-loop system is proved.

### 164 **3.1 Intelligent proportional-integral Control**

For a general single input single output (SISO) nonlinear system, an ultra-local modelwhich is defined as follows can be used to define its corresponding dynamics

$$\mathbf{y}^{(n)} = F + \boldsymbol{\alpha} \cdot \boldsymbol{u} \,, \tag{7}$$

168 where  $n \ge l$  is the derivative order of the output *y*, and *u* is the input,  $\alpha$  is the input 169 gain, *F* is the lumped unknown dynamics (LUD) disturbance. If n = 1, a first-order 170 system can be selected to describe the dynamics of the controlled system.

171 If F and  $\alpha$  are well-known items, an iPI control can be proposed as

172 
$$u = \frac{1}{\alpha} (-F + \dot{y}^* + k_p e + k_i \int e dt), \qquad (8)$$

173 where  $e = y^* - y$  is the output error and  $y^*$  is the desired reference. Substituting 174 equation (8) into (7), the error equation can be deduced as follows

$$\dot{e} + k_p e + k_i \int e dt = 0. \tag{9}$$

176 The steady error dynamics of this closed loop is determined by the parameters  $k_p$  and 177  $k_i$ , whose values can be selected according to the Hurwitz criterion.

178 Let us now fit this model to our electro-mechanical system. Combining the equations179 (2,4) and (5), the active and reactive powers are calculated as follows

180  

$$\begin{cases}
P_{s} = \frac{3}{2}(u_{sd}i_{sd} + u_{sq}i_{sq}) = \frac{3}{2}\frac{L_{m}}{L_{s}}u_{sq}i_{rq} \\
Q_{s} = \frac{3}{2}(u_{sq}i_{sd} - u_{sd}i_{sq}) = \frac{3}{2}\frac{\psi - i_{rd}L_{m}}{L_{s}}u_{sq},
\end{cases}$$
(10)

181 The active and reactive powers are decoupled and they are only related to d axis or q182 axis rotor current. The same iPI controller is retrieved for the active and reactive 183 powers.



184

167

185 **Fig. 3.** Two loop vector control for DFIG (PI for Current, PI for Power)

Here, we only explain for the active power case. According to equation (10), thedynamic active power equation can be approximately written as

188

192

$$\dot{P}_s = F + \alpha \cdot \dot{i}_{ra},\tag{11}$$

189 where *F* is a disturbance related to turbulent wind and other factors such as *d* axis 190 coupled current, and  $\alpha = 1.5 L_m u_{sa} / L_s$ .

- 191 The power error is defined as
  - $e = P_{opt} P_s, \tag{12}$

193 where  $P_{opt}$  is the optimal power obtained on the power chart.

194 In this paper, the estimated disturbance  $\hat{F}$  will be obtained using an extended state 195 observer (ESO) method [25]. According to the ESO method, a second observer is 196 introduced

197  

$$\begin{cases}
 e_1 = z_1 - P_s \\
 \dot{z}_1 = z_2 - \beta_1 e_1 + \alpha u \\
 \dot{z}_2 = -\beta_2 |e_1|^{1/2} sign(e_1)', \\
 \hat{F} = z_2
 \end{bmatrix}$$
(13)

198 where  $\beta_1$ ,  $\beta_2$  are constant.  $e_1$  is the estimation error of ESO.  $z_1$  and  $z_2$  are the 199 intermediate states. In the ESO framework,  $z_2$  represents the estimation produced by 200 the observer, we thus have  $z_2 = \hat{F}$ . Therefore, for the first order system (18), the 201 following relatively simple intelligent PI (iPI) control can be proposed to achieve 202 optimal power tracking 203  $\dot{i} = \frac{1}{2}(-\hat{F} + \dot{P} + k_1)e_1 + k_2 + k_3 \int e_1 d\tau$  (14)

$$i_{rq} = \frac{1}{\alpha} \left( -\hat{F} + \dot{P}_{opt} + k_p \cdot e + k_i \cdot \int e d\tau \right).$$
(14)

From the ESO equations, the estimated error exists and is defined as  $\tilde{F} = F - \hat{F}$ .

From reference [18], one has generally  $\|\tilde{F}\| \le f_m$  with  $f_m$  an upper bound value.

206 Substituting equation (14) into (11), the error equation is deduced as

$$\dot{e} + k_p \cdot e + k_i \cdot \int e dt + \tilde{F} = 0.$$
(15)

208 Applying the Laplace transform to equation (15), we obtain:

209 
$$(s + k_p + k_i / s)E(s) + \tilde{F}(s) - \tilde{F}(0^+) = 0.$$
(16)

210 According to final value theorem, the steady error can be calculated as

211 
$$e(t_{\infty}) = \lim_{t \to \infty} e(t) = \lim_{s \to 0} \frac{s^2}{s^2 + k_p \cdot s + k_i} (\tilde{F}(0^+) - \tilde{F}(s)).$$
(17)

212 Since  $\|\tilde{F}\|$  is bounded and  $k_p$  and  $k_i$  are selected as Hurwitz polynomial parameters, 213 the steady error  $e(t_{\infty})$  is ensured to tend to zero. According to the steady error

- equation (17), the performance of iPI controller depends on the gains  $k_p$  and  $k_i$  and on
- 215 the estimated value of F. If the result of the observer is not accurate, this method will
- 216 be ineffective. In addition, the measurement noise of power will also weaken the
- 217 performance because estimation error will increase, especially in presence of
- 218 high-frequency noise.

# 219 **3.2 Intelligent Proportional-integral Sliding Mode Control**

In this part, we add an extra input to compensate the estimation error and measurement noise. The structure of this iPISMC control is shown in Fig 4.

The extra input is denoted by  $u_e$ . The final intelligent proportional and integral sliding mode controller (iPISMC) can be defined as

224 
$$\dot{i}_{rq} = \frac{1}{\alpha} \left( -\hat{F} + \dot{P}_{opt}^* + k_p \cdot \mathbf{e} + k_i \cdot \int e d\tau \right) + u_e.$$
(18)

The structure of this iPISMC control is shown in Fig. 4. By substituting equation (18)in equation (11), the closed-loop error is given by:

235

236

 $\dot{e} + k_p \cdot e + k_i \cdot \int e d\tau + \alpha \cdot u_e + \tilde{F} = 0.$ <sup>(19)</sup>

228 Define  $x_1$  and  $x_2$  as follows:

229  $\begin{cases} x_1 = \int e d\tau \\ x_2 = e \end{cases}, \tag{20}$ 

230 The state-space equations can be obtained as

231 
$$\begin{cases} \dot{x}_1 = x_2 \\ \dot{x}_2 = -k_p * x_2 - k_i * x_1 - \alpha * u_e - \tilde{F} \end{cases}$$
 (21)

Therefore, the extra input  $u_e$  is designed to compensate the disturbance. According to the sliding mode control framework, a switching function *S* is defined as

234 
$$S = x_2 + c * x_1$$
. (22)



237 The derivative of equation (22) is

238

$$S = -k_i * x_1 + (c - k_p) x_2 - \alpha * u_e - F.$$
(23)

In order to ensure the stability of closed-loop system, the input should be selected so that state trajectories are confined to the sliding hyper surface. The extra input  $u_e$  is composed by two parts:

• equivalent control signal  $u_1$  which ensures the ideal sliding mode condition (S=0),

• correction control signal  $u_2$  which reduces the chattering effects.

244 The extra input is

 $u_e = u_1 + u_2. (24)$ 

246 Considering that  $\tilde{F}$  is unknown in equation (23) and the ideal sliding mode condition, 247  $u_1$  is calculated by replacing  $\tilde{F}$  with  $f_m$  as following

248 
$$u_1 = \frac{1}{\alpha} (-k_i * x_1 + (c - k_p) x_2 - f_m).$$
(25)

249 In order to reduce the chattering effects,  $u_2$  is selected as

250 
$$u_2 = \frac{1}{\alpha} (\eta_1 * sat(S, \varepsilon) + \eta_2 * S),$$
 (26)

251 where 
$$sat(S,\varepsilon) = \begin{cases} 1 & , S > \varepsilon \\ S / \varepsilon & , \|S\| \le \varepsilon \text{ and } \eta_1 > 0, \eta_2 > 0, \varepsilon > 0 \\ -1 & , S < -\varepsilon \end{cases}$$

252 The input  $i_{rq}$  can be rewritten as follows

253  
$$i_{rq}(t) = \frac{1}{\alpha} (-\hat{F} + \dot{P}_{opt}^* + k_p \cdot e + k_i \cdot \int ed_\tau) + u_e$$
$$= \frac{1}{\alpha} (-\hat{F} + \dot{P}_{opt}^* + \eta_1 \cdot sat(S, \varepsilon) + \eta_2 \cdot S + c \cdot e)$$
(27)

### 254 **3.3 Stability analysis**

- 255 Define the following Lyapunov function as
- 256  $V = \frac{1}{2}S^2$ . (28)
- 257 The derivative of equation (28) is

258  
$$\dot{V} = S\dot{S} = S(\dot{x}_2 + c * \dot{x}_1) = -S(\eta_1 \cdot sat(S, \varepsilon) + \tilde{F} - f_m) - \eta_2 S^2.$$
(29)

259 If  $S < -\varepsilon$ , the boundedness of  $\tilde{F}$  is sufficient to ensure that 260  $\dot{V} = -S(-\eta_1 + \tilde{F} - f_m) - \eta_2 S^2 < 0$ . If  $S > \varepsilon$ ,  $\dot{V} = -S(\eta_1 + \tilde{F} - f_m) - \eta_2 S^2$ . According to 261 the boundedness of  $\tilde{F}$ , it is ensured that  $\dot{V} < 0$  if one has  $\eta_1 > 2f_m$ .

262 If  $|S| < \varepsilon$ , we obtain

263 
$$\dot{V} = -S(\eta_1 S / \varepsilon + \tilde{F} - f_m) - \eta_2 S^2.$$
(30).

264 In order to ensure the negative right-side term, the condition is

265 
$$\frac{-S(\tilde{F} - f_m) - (\eta_2 + \eta_1 / \varepsilon)S^2 < 0}{|(\tilde{F} - f_m)| < (\eta_2 + \eta_1 / \varepsilon)|S|}.$$
(31)

266 Again, given the boundedness of  $\tilde{F}_{and} |S| < \varepsilon$ , the condition is  $(\eta_1 + \varepsilon \eta_2) > 2f_m$ .

267 In summary, the conditions needed to ensure stability of closed loop system are

268  $\eta_1 > 2f_m \text{ and } \eta_2 > 0, \varepsilon > 0$ .

The reactive power is related to the d axis rotor current. Usually, the reference of reactive power is set to zero. The structure of reactive power controller can be chosen the same as the above iPISMC control for the active power. Its stability can be proved likewise.

# 273 4. Simulation Results

To validate the effectiveness of this proposed iPISMC control, we tested it on the co-simulation platform of Matlab/Simulink and FAST. The main parameters for computer simulations are shown in table 1. The wind turbine model originates in FAST platform which is developed by the National Renewable Energy Laboratory (NREL) [26]. The detailed model and parameters of 5MW DFIG are selected from reference [27].

In this paper, we only investigate three controllers: PI, iPI and iPISMC. The parameters  $k_p = 5 \times 10^{-5}$  and  $k_i = 2.5 \times 10^{-4}$  are set to the same values for all three methods. This is justified by the fact that iPI is a wrapper for PI and iPISMC is a wrapper for iPI. Moreover, these parameters are optimized using the pole placement method.

### 285 **4.1 Stochastic wind**

In order to demonstrate the performance in more realistic conditions, a stochastic wind has been utilized and the results are shown in Fig. 5. The stochastic wind speed

 $\frac{288}{289}$ 

**Table 1.** The main parameters of wind turbine system

Parameter Description	Value
Rated Power	5 MW
Rotor Radius	63 m
Gear Box Ratio	97
Moment of inertia	$4.38E + 07N.m^2$
Frequency	50Hz
Number of Pole pairs	3
Stator resistance	$1.552m\Omega$
Stator Leakage inductance	1.2721mH
Rotor resistance	$1.446m\Omega$
Rotor Leakage inductance	1.1194mH
Mutual inductance	5.5182mH



295 mean value is 8.0 m/s. The chosen turbulence model is an international 296 electrotechnical commission (IEC) standard Kaimal model produced by TurbSim 297 software. A realization of that stochastic process is given in Fig. 5(a). Its 298 corresponding tracking performance results are illustrated in Fig. 5 (c-d). It can be 299 noticed clearly from Fig. 5(d) that the proposed iPISMC ensures the best optimal 300 power tracking performance compared with the classical PI and iPI methods.

301 302

#### Table 3. The mean power under PI, iPI and iPISMC

(Mean value=8m/s, IEC standard Kaimal model)										
Criterions	$k_p = 5.0 \times 10^{-5}$ $k_i = 2.5 \times 10^{-4}$			$k_p = 1.9 \times 10^{-5}$ $k_i = 9.3 \times 10^{-4}$			$k_p = 1.0 \times 10^{-4}$ $k_i = 5.1 \times 10^{-4}$			
	PI	iPI	iPISMC	PI	iPI	iPISMC	PI	iPI	iPISMC	
Mean power (value $\times 10^5$ )	1.9759	1.9801	1.9834	1.9586	1.9727	1.9834	1.9802	1.982	1.9834	
Mean error of power error (value ×10 <sup>5</sup> )	0.4705	0.2268	0.0007	1.3228	0.6453	0.0007	0.2218	0.1089	0.0007	
Variance of power error (value ×10 <sup>11</sup> )	1.6456	0.7441	0.0102	4.6253	2.1684	0.0102	0.7856	0.3451	0.0102	

303 A complementary numerical analysis is provided by table 3. Different controller 304 parameters are selected and tested. Comparing PI with iPI under the same conditions,

305 the mean power using iPI is bigger than that of PI, while the mean error and variance



Fig. 6. Simulation results with  $R_r = 100\%$  120% 150% and 180%

311 are smaller. With the same parameters  $k_p$  and  $k_i$ , the mean power obtained when using iPISMC is bigger than that of other methods. Furthermore, the values of mean error 312 313 and variance reflect iPISMC efficiency. It shows that iPISMC also outperforms PI and

314 iPI controllers regarding this criterion.

#### 315 4.2 Step wind with parameter variations

316 In order to test the influence of DFIG parameter variations on the performances of 317 the proposed iPISMC, different conditions with parameter variation of resistances 318 and mutual inductance have been tested and the corresponding results are reported in Fig. 6 - 8. For instance, the resistance is sensitive to the temperature which 319 320 changes gradually with respect to ambient temperature. Consequently, the rotor 321 resistances and mutual inductance are considered and tested.

322 Fig. 6-8 respectively shows the results of generator rotor speed, power and its tracking 323 error under the variations of resistances  $a \cdot R_r$ ,  $a \cdot R_s$  and the mutual inductance with

324 
$$a \cdot L_m$$
 with  $a = \{1; 1.2; 1.5; 1.8\}$ .

325 From the figures, the power errors converge to zero rapidly. Parameter variations have no significant influence on output rotor speed or power. From these results, one can 326

327 notice that our iPISMC is robust and able to reject the influences of the variations of





337338

(c) Output power of generator (d) Output power error of generator **Fig. 8.** Simulation results with  $L_m = 100\%$  120% 150% and 180%

# 339 **5.** Conclusion

In this paper, an intelligent proportional-integral sliding mode control for direct power control of variable speed-constant frequency wind turbine system is presented. This controller consists in two nested controllers: an intelligent proportional integral controller enhanced by a sliding mode compensated controller. In order to demonstrate its performance, the controller is tested in two different cases which include stochastic wind and parameter variations.

346 Under stochastic wind turbulences, the average output error of iPISMC is 347 significantly smaller than that of PI or iPI. Moreover, iPISMC is not sensitive to 348 unpredictable parameter variations. This also tends to show that iPISMC can be 349 employed when these parameters are ill-known. Consequently, iPISMC is well suited 350 for DFIG wind turbine robust control in practical situations.

351

# 352 Acknowledgments

353 This work is supported by National Nature Science Foundation of China(61304077,

354 61203115), Jiangsu Province under Grant(BK2013075), by the Chinese Ministry of

- 355 Education Project of Humanities and Social Sciences (13YJCZH171), by the Funding
- 356 of Jiangsu Innovation Program for Graduate Education(Grant No.KYLX 0377)
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